#### Comparing Anisotropic Output-Based Grid Adaptation Methods by Decomposition

#### Mike Park NASA Langley Research Center

Adrien Loseille INRIA Paris-Rocquencourt

Joshua Krakos and Todd Michal The Boeing Company

## Motivation

 Mesh generation and adaptivity continue to be significant bottlenecks in the CFD workflow, and very little government investment has been targeted in these areas.
 – CFD Vision 2030 Study (NASA-CR-2014-218178)

## Approach

- Create and sustain collaboration in solution adaptive research with a goal of addressing the CFD Vision 2030 goals of automation, uncertainty quantification, and robustness
  - Leverage research groups across different organizations around the world

### **Output-Based Adaptation Process**



• Use error estimation and grid adaptation to reduce the requirements on the initial grid

## **Output-Based Adaptation Process**



- There are may elements involved and they need to be right for the entire procedure to converge
  - To ensure correctness, examine elements individually and as a whole

## **Output-Based Adaptation Process**



- Today we will focus on grid adaptation mechanics for triangular and tetrahedral grids
  - In the context of specified spacing field and solution error estimation procedures

## Metric

- Examine metric-based approaches to unstructured grid adaption
- Metric is a 3x3 (or 2x2 in 2D) matrix to define an orthogonal basis and spacing in each basis vector direction
  - Stored at each vertex in the mesh and interpolated as needed
- Edge lengths are computed in the metric
   An ideal grid has all unit-length edges in metric

## **3D Prescribed Metric Field**

- Very simple metric field, but still illustrative
- Defined in a unit square
- X-spacing and Y-spacing is constant 0.1
- Z-spacing varies linearly from 0.1 at the top and bottom to 0.001 at the center of the square
- Introduction of the adaptive mechanics













# refine/two

#### Edge-based algorithm using only insertion and collapse operators



## **EPIC-IC**

#### Edge-based algorithm using only insertion and collapse operators



## **EPIC-ICS**

#### Edge-based algorithm using insertion, collapse

and swap



## **EPIC-ICSM**

Edge-based with insertion, collapse, swap, and node movement



## refine/one

Edge-based with insertion, collapse, swap, and node movement



#### Edge length in metric

## Feflo.a

Cavity-based with insertion, collapse, swap, and node movement



## Laminar Delta Wing

- International Workshop on High-Order CFD Methods test case
  - Mach 0.3, 12.5° angle of attack, and 4000
     Reynolds number based on mean chord



## Laminar Delta Wing

- International Workshop on High-Order CFD Methods (HIOCFD) test case
  - Mach 0.3, 12.5° angle of attack, and 4000
     Reynolds number based on mean chord
- Venditti metric with refine/one, refine/two, Feflo.a, and EPIC-ICS
- Multiple adaptations at series of increasing complexity (size request)



## Laminar Delta Wing Drag



Cd\_ref-Cd

0.1 24



# EPIC-ICS Delta Wing Grid



## 2D Flat Plate

- Available on the Turbulence Modeling Resource (TMR) website
- Examined in a SciTech 2015 special session
- Not a production capability yet
  - Plagued by iterative convergence and Hessian reconstruction (error estimation) issues
  - Grid adaption mechanics are available for this case (1000-1 aspect ratio for solver robustness)



## 2D Flat Plate Drag Convergence

Tracking of CFL3D drag coincidental



## 2D Flat Plate Velocity Profile

Overlays structured grid result



#### 2D Flat Plate Skin Friction

• Better resolves leading edge singularity



#### 2D Flat Plate Skin Friction

• Slight noise in skin friction



#### 2D Flat Plate Grid

• Initial grid is uniform and isotropic



## 2D Flat Plate Grid

• Anisotropic refinement near the plate



#### 2D Flat Plate Z-Scaled Grid

• 1000-1 refinement appears isotropic



## Summary

- Simple specified metric illustrates the properties of each adaptive mechanics tool

   Statistics in paper verify histogram observations
- Nominally second order adaptive finite volume scheme competitive with HIOCFD-2 results in terms of drag error per degree of freedom for multiple grid adaptation tools
  - Avenue for collaboration on solver technology, error estimation, and tetrahedral grid adaptation

## Summary

- While not a routine capability, turbulent 2D flat plate shows promise
  - Hampered by solver robustness for under resolved flow features and high aspect ratio unstructured grids
  - Experimental solver technology and discretization improvements very helpful
  - Error estimation (Hessian reconstruction, particularly on boundaries) should be improved to produce smoother metric variation

## In the Paper

- Compiled statistics to quantify edge length histograms
- 2D specified metric field
- Diamond airfoil Mach 2 drag adaptation
  - 2D triangles and 3D extruded to unit span tetrahedra
  - Comparison of Venditti and INRIA Optimal-Goal metrics
- Description of the error estimate procedures, Hessian reconstruction (boundary), and gradation control

## Thank you!

- Turbulence Modeling Resource and High Order CFD Workshop websites invaluable
  - Big thank you to the people supporting these community resources

## Future work

- SciTech 2016 paper and beyond (summarized in the paper)
  - Metric aligned grid elements
  - Curved boundaries and geometry access
  - Parallel execution
  - Error estimation
  - 2D and 3D RANS with turbulence modeling
  - Time-accurate simulation
- Engage researchers in a sustained effort and disseminate findings (paper, website, workshop)